

Thus, we have

(6)

$$F = 4mk^2 \left(\frac{1}{d} + \frac{\epsilon \cos \theta}{d} - \frac{\epsilon \cos \theta}{d} \right) \cdot \frac{1}{r^2}$$
$$= \frac{4mk^2}{d} \cdot \frac{1}{r^2}$$

Which is an inverse square force.

Inverse Square Force \Rightarrow Conic Trajectory

(1)

Newton's Law of Universal Gravitation

$$F = G \frac{Mm}{r^2}$$

G = gravitational constant

M = mass of object

m = mass of other object

r = distance between

Let's write $C = GM$ and choose the object with mass M to be at the origin.

Comparing this Force with our previous formula, we get

(2)

$$F = m \left(\frac{4k^2}{r^3} - \frac{d^2 r}{dt^2} \right) = C \frac{m}{r^2}$$

$$\frac{4k^2}{r^3} - \frac{d^2 r}{dt^2} = \frac{C}{r^2} \quad (*)$$

Let $r = f(\theta)$ & $g(\theta) = \frac{1}{f(\theta)}$, so that

$$r = g(\theta)^{-1}$$

Then

$$\begin{aligned} \frac{dr}{dt} &= -\left(g(\theta)\right)^{-2} \frac{d}{dt}(g(\theta)) = -\left(g(\theta)\right)^{-2} g'(\theta) \frac{d\theta}{dt} \\ &= -\left(g(\theta)\right)^{-2} g'(\theta) \left(\frac{2k}{r^2}\right) = -g(\theta)^{-2} g'(\theta) (2kg(\theta)^2) \\ &= -2kg'(\theta) \end{aligned}$$

and one more derivative

$$\frac{d^2 r}{dt^2} = -2kg''(\theta) \cdot \frac{d\theta}{dt} = -2kg''(\theta) \cdot \frac{2k}{r^2} = -4kg''(\theta) \cdot \frac{1}{r^2}$$

Plugging this into (*) gives

$$\left[\frac{4k^2}{r^3} + 4k^2 g''(\theta) \cdot \frac{1}{r^2} = \frac{C}{r^2} \right] \cdot \frac{r^2}{4k^2} \quad (r \neq 0) \quad [3]$$

$$\Rightarrow \frac{1}{r} + g''(\theta) = \frac{C}{4k^2}$$

$$\Rightarrow g''(\theta) + g(\theta) = \frac{C}{4k^2}$$

Letting $g(\theta) = h(\theta) + \frac{C}{4k^2}$, we get

$$h''(\theta) + h(\theta) = 0 \Leftrightarrow h''(\theta) = -h(\theta)$$

The two functions which satisfy this are

$$\sin \theta \text{ & } \cos \theta$$

The most general solution is then

$$h(\theta) = A \sin \theta + B \cos \theta$$

$$\Rightarrow g(\theta) = A \sin \theta + B \cos \theta + \frac{C}{4k^2}$$

At this point, we can forget that θ is a function of t , since we're looking for the shape of the trajectory.

(4)

Unless other forces act on our particle,
 there will be a point of "closest approach" for
 P to the center of the force. This means
 that $f(\theta)$ has a local minimum, which means
 $g(\theta)$ has a local maximum. By rotating the
 axes if necessary, we can assume that this is
 at $\theta=0$. This means that $g'(0)=0$

$$g'(\theta) = A \cos \theta - B \sin \theta$$

$$g'(0) = A = 0$$

$$\text{So, } g'(\theta) = -B \sin \theta \Rightarrow g''(\theta) = -B \cos \theta$$

Now, if $B < 0 \Rightarrow -B > 0 \Rightarrow g''(0) = -B \cos 0 = -B > 0$
 $\Rightarrow g$ has a local minimum at $\theta=0$, a contradiction.

Thus $B \geq 0$

$$\Rightarrow g(\theta) = B \cos \theta + \frac{C}{4K^2}$$

$$\Rightarrow f(\theta) = \frac{1}{B \cos \theta + \frac{C}{4K^2}} = \frac{\frac{4K^2}{C}}{1 + \frac{4K^2B}{C} \cos \theta}$$

$$\text{So, } d = \frac{4\kappa^2}{C} \quad \& \quad \varepsilon = \frac{4\kappa^2 B}{C}$$

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fitting this with

Elliptical orbit

$$r = f(\theta) = \frac{d}{1 + \varepsilon \cos \theta}$$

⇒ Kepler's First Law
(Only closed orbits are ellipses)

$L = 2d$ is the latus rectum.

a = semimajor axis b = semiminor axis

T = period of the orbit

$$\Rightarrow \kappa = \frac{\pi ab}{T} \quad \& \quad L = \frac{2b^2}{a}$$

$$\Rightarrow C = \frac{4\kappa^2}{d} = \frac{8\kappa^2}{L}$$

$$\Rightarrow F = \frac{Cm}{r^2} = \frac{8mk^2}{L} \cdot \frac{1}{r^2} = \cancel{8m} \cdot \frac{9}{\cancel{2b}^2} \cdot \frac{\pi^2 a^2 b^2}{T^2} \frac{1}{r^2}$$

$$\Rightarrow F = \frac{4m\pi^2 a^3}{T^2} \frac{1}{r^2} = GmM \cdot \frac{1}{r^2}$$

$$\Rightarrow \frac{4\pi^2 a^3}{T^2} = GM \iff \boxed{\frac{a^3}{T^2} = \frac{GM}{4\pi^2}}$$

Kepler's 3rd Law.